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OPSECRET FII AFSSO ATIC TO AFSSO USAF TOP SECRET FROM: ND-0002-9. THIS IS A MOTOR DROME

25X1A

WESSAGE TO TSO FISHER FOR PASSING TO

25X1A

AFTER IT HAS BEEN SHOWN TO GEN VATSON AND/OR FROM TCO POTTER.

25X1A

THE FLIGHT TEST RESULTS SUBMITTED BY

HAVE BEEN REVIEWED AND THE FOLLOWING

COMMENTS ARE SUBMITTEDS

GEN WALSH.

- THERE ARE NO SOVIET FIGHTER AIRCHAFT CAPABLE OF INTERCEPTING THE U-2 AIRPLANE WITHOUT THE USE OF AIR-TO-AIR MISSILES. IT IS ESTIMATED THAT IR, DEAM RIDER, AND SEMI-ACTIVE HOMING MISSILES ARE CURRENTLY AVAILABLE FOR USE BY SOVIET FIGHTERS.
- (B) OF THE KNOWN AIRCRAFT CURRENTLY IN ORERATION WITH THE SOVIET AIR FORCES ONLY FISHPOT AND FITTER ARE ESTIMATED TO HAVE A CO-ALTITUDE ATTACK CAPABILITY AGAINST THE U-22 HOWEVER, THIS CAN BE ACCOMPLISHED ONLY IN A DYNAMIC-CLIMB CONDITION WITH CONTINUALLY VARYING ALTITUDE AND SPEED, AND WITH LOSS OF CONTROL EFFECTIVENESS AND MANEUVERABILITY AT OR NEAR THE PEAK POINT. THE

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DYNAMIC CLIMB CAPABILITIES OF FISHPOT AND FITTER ARE CONSIDERABLY LESS THAN THE F-104 CAPABILITY, ESPECIALLY WHEN MISSILES ARE BEING CARRIED. COMPUTER DATA INDICATE THAT FITTER COULD NOT EXCEED 75,000 FEET IN THE CLEAN CONFIGURATION UNDER OPTIMUM CONDITIONS.

- ALTITUDES IS PROBABLY CAUSED BY THE DECREASED AMBIENT
 PRESSURES AT THESE ALTITUDES. ADEQUATE PRESSURIZATION OF THE AI
 RADAR COMPONENTS WOULD SIGNIFICANTLY HELP. FISHPOT IS
 ESTIMATED TO HAVE A SEARCH TRACK RADAL AND IR SIGHTING
 SYSTEM SIMILAR TO THE F-104, GIVING AN ALL WEATHER
 CAPABILITY. IT IS ESTIMATED THAT THE FISHPOT RADAR WOULD
 PROBABLY SUFFER SUCH DIFFICULTIES UNDER SIMILAR CONDITIONS,
 HOWEVER, ADEQUATELY PRESSURIZING THE AI EQUIPMENT IS NOT TOO
 DIFFICULT TO ACCOMPLISH.
- (D) IT IS ESTIMATED THAT FITTER IS EQUIPPED WITH INFRA-RED ANGLE TRACK AND RANGE ONLY RADAR. THIS IN CONJUNCTION WITH A CLOSE CONTROL GCI ENVIRONMENT PROVIDES A CAPABILITY FOR A NON-VISUAL CLEAR AIR MASS INTERCEPTION. IT IS BELIEVED THAT IR SIGHTING SYSTEMS ARE NOT SUBJECT

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N/A

TOP SEGNET

TO THE SAME ALTITUDE DEGRADATION PENALTIES THAT AFFECT AT RADAR SYSTEMS. THE EGLIN PLIGHT TEST RESULTS DO NOT SHOW THE USZ AND EFFECTIVENESS OF IR SIGHTING SYSTEMS AT ALTITUDE.

- (E) CONSIDERATION OF THE USE OF DYNAMIC CLIMB
- AIRPLANE MUST BE ACCOMPLISHED PRIOR TO OR DURING THE INITIAL PART OF THE CLIMB.
- (2) DISTANCE FROM THE INITIAL ACCELERATION POINT TO PEAK OUT POINT IS CONSIDERABLE AND GENERALLY PRECLUDES THE USE OF THE AIRCRAFT'S TRACKING RADAR UNTIL THE AIRCRAFT IS ABOVE ITS CRITICAL MANEUVERING ALTITUDE.
- ALTITUDE, ON TARGET, AND IN A HISSILE FIRING POSITION SIMULTANEOUSLY TO SUCCESSFULLY PERFORM ITS INTERCEPTION HISSION.

BASED ON THESE CONSIDERATIONS OF THE FLIGHT TEST RESULTS, IT IS FELT THE CHANCES OF SUCCESS OF A DYNAMIC CLIMB OF FISHPOT AND FITTER IN A CO-ALTITUDE ATTACK ON THE U-2

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IS RATHER POOR. HOWEVER, IT IS FELT THAT IT IS NOT NECESSARY TO ATTACK ONLY AT THE CO-ALTITUDE POSITION.

IT IS POSSIBLE TO LAUNCH THE MISSILES AT ANY TIME THAT LOCK ON IS ACQUIRED, PROBABLY FROM AS MUCH AS 10-15,000 FEET BELOW THE U-2. THE ADVANTAGES OF LAUNCHING AT SUCH A POINT ARE THAT THE ATTACKING AIRPLANE HAS MORE MANEUVER MARGIN REMAINING FOR FINAL POSITIONING THAN HE WOULD HAVE UPON REACHING CO-ALTITUDE, AND THAT THE RELIABILITY OF THE FIRE CONTROL SYSTEM IS BETTER AT LOWER ALTITUDES. IT IS NOT APPARENT FROM THE TEST RESULTS WHY THE F-104 WAS NOT UTILIZED IN THIS MATTER.

(F) CONSIDERATION OF THE SOVIET ALL-VEATHER FIGHTERS
IN ADDITION TO FISHPOT THAT WOULD BE OPERATIONAL FOR THE TIME
PERIOD CONSIDERED, SHOWS THAT FLASHLIGHT A AND THE
ESTIMATED 1959 ALL-WEATHER FIGHTER WOULD BE AVAILABLE FOR
ATTACKS AGAINST THE U-2. AS PUBLISHED IN ATIC STUDIES,
FLASHLIGHT A CAN BE EMPLOYED IN NON-CO-ALTITUDE (PULL UP)
TYPE ATTACKS AT SUB-SONIC SPEEDS. THE 1959 ALL-WEATHER
FIGHTER PROVIDES AN IMPROVED SERIES OPERATING IN THE
MACH 1.5 SPEED REGIME, AND HAVING A COMBAT CEILING OF

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60,300 FEET. AGAIN, THIS AIRPLANE CAN BE EMPLOYED IN NON-CO-ALTITUDE (PULL UP) TYPE ATTACKS WHICH WERE PROVEN SO SUCCESSFULLY IN THE F-102 INTERCEPTIONS OF THE U-2.

RECEIVED INDICATING THE POSSIBLE ADDITION OF ROCKET
HOTOR PROPULSION UNITS ON SOME FRESCO AND FLASHLIGHT
FIGHTERS. IT IS BELIEVED THAT THE INSTALLATION OF THESE
ROCKET HOTORS IS NOT A GENERAL BETRO-FIT MODIFICATION PROGRAM.
IT IS POSSIBLE THIS ROCKET MOTOR INSTALLATION IS INTENDED
FOR USE IN ATTACKS AGAINST SELECTED HIGH-ALTITUDE TARGETS
SUCH AS THE U-2. AS WAS SHOWN IN THE FLIGHT TEST RESULTS
ADDITIONAL THRUST DURING A PULL-UP MANEUVER WOULD DE
USEFUL SO THAT CONTROL CAN DE MAINTAINED WHILE ATTEMPTING
LOCK ON AND MISSILE FIRING.

(H) DUE TO THE DIFFICULTY IN SETTING DOWN IN DETAIL

ALL OF THE RAMIFICATIONS OF THE EGLIN SPECIAL FLIGHT TEST

RESULTS ON ESTIMATED SOVIET INTERCEPTOR AIRCRAFT

CAPABILITIES, IT IS SUGGESTED THAT A MEETING BE HELD AT ATIC WITH YOUR ORGANIZATION AS SOON AS POSSIBLE TO FULLY EXPLOIT THESE RESULTS.

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